



FY25 Strategic University Research Partnership (SURP)

Multi-Objective and Multi-Scale Planning for Long-Range Planetary Rover Autonomy

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Objectives: This work develops and validates onboard planning algorithms to enable robotic systems to consider multiple objectives while operating autonomously. We use Endurance as the mission use case, but our work is broadly applicable to autonomous rovers, spacecraft, and aerial vehicles. Our methods use a holistic understanding of the mission environment that includes science objectives, operation constraints, localization uncertainty, and navigation safety. We research multi-objective planning methods and evaluate them using real and relevant orbital of the Moon and Earth.

Background: The primary purpose of planetary explorers is to collect scientific information. Most decisions are made by humans balancing science goals with rover operations (power, communication, safety, etc.). Extensive ground-in-the-loop planning can become a bottleneck that reduces productivity (Gaines et al. 2017). There has been immense progress in rover autonomy to improve operations efficiency (Verma et al. 2023). Yet existing rover autonomy software cannot be directly repurposed for upcoming lunar missions. Lunar rovers will drive faster and farther at a tempo that precludes continuous ground guidance, with planning further complicated by dynamic illumination, temperature, and communication limits. Our work addresses these challenges for the Endurance mission concept: a ~2000 km traverse requiring a complex balance of science goals against constraints (Keane et al. 2021).

Approach and Results: The core of this work is a framework for adapting standard single-objective path planning algorithms, such as A*, for multi-objective tasks. We achieve this by using multi-objective decomposition, a technique that transforms a complex multi-objective problem into simpler, single-objective ones (Miettinen, 1998). We adapt four path planners to incorporate these decomposition methods: A*, MCTS, fixed-horizon search, and greedy search. Our approach compares three existing decomposition methods—Weighted Sum, Tchebychev, and Boundary Intersection (BI)—and introduces a novel method: a Euclidean Distance decomposition that finds the solution closest to an ideal "reference point" in the objective space (Fig. 1).

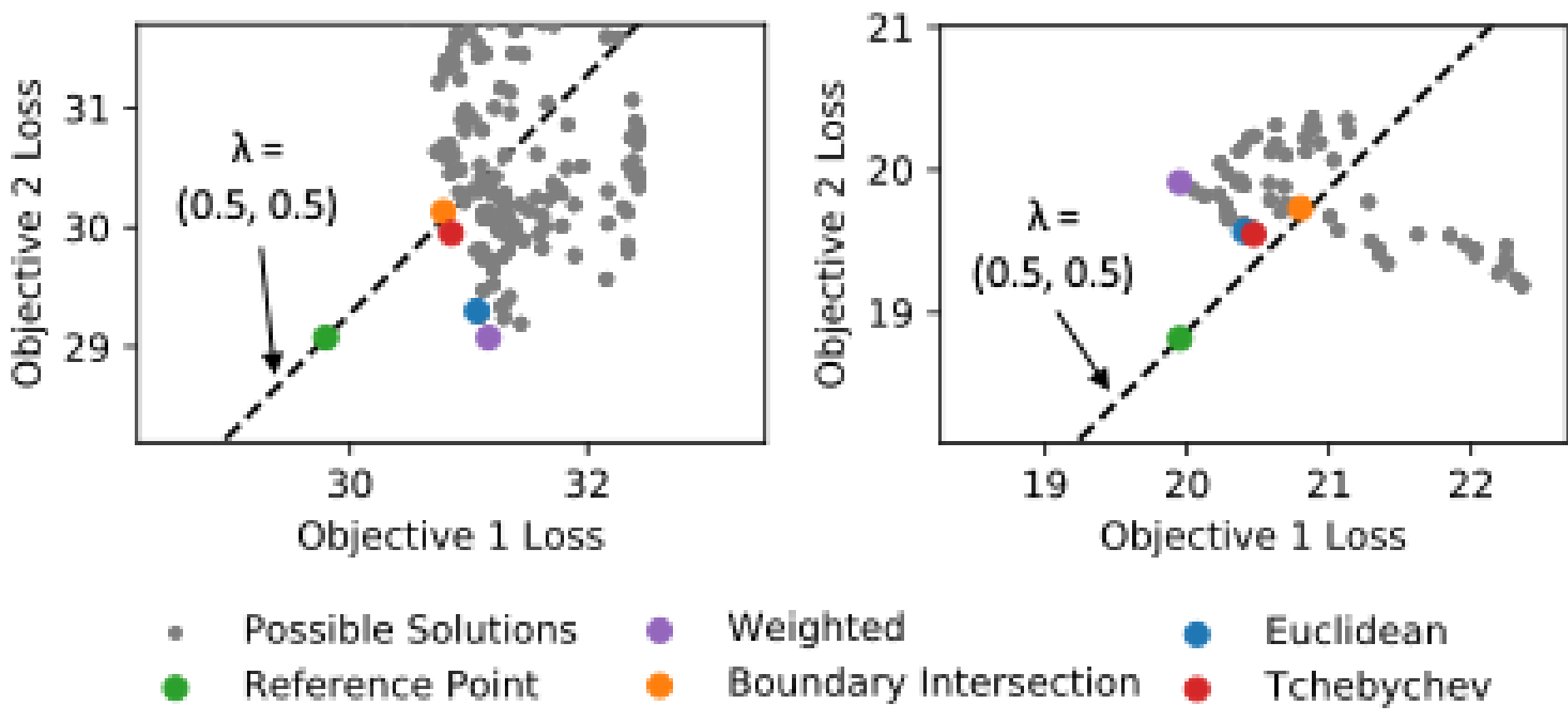


Fig. 1. Potential solutions for a path over two objectives. The weighted method of decomposition has varied and unpredictable behavior in objective space, while other methods are explainable and consistent.

We test the algorithms in simulations using two realistic rover exploration scenarios and real-world data (Fig 2). For the Lunar Rover Scenario, which is based on the Endurance mission concept (Keane et al. 2021), the rover must balance four objectives: scientific interest, lighting conditions, safety, and "keep out" regions; we use data from the USGS

Unified Geologic Map of the Moon (Fortezzo et al. 2020) and LOLA elevation models (Barker et al. 2016). The Terrestrial Rover Scenario involves a rover exploring a geologically significant area in Cuprite, NV and balancing three objectives: scientific interest (based on an evolving entropy map from Candela et al. 2020), lighting conditions, and safety.

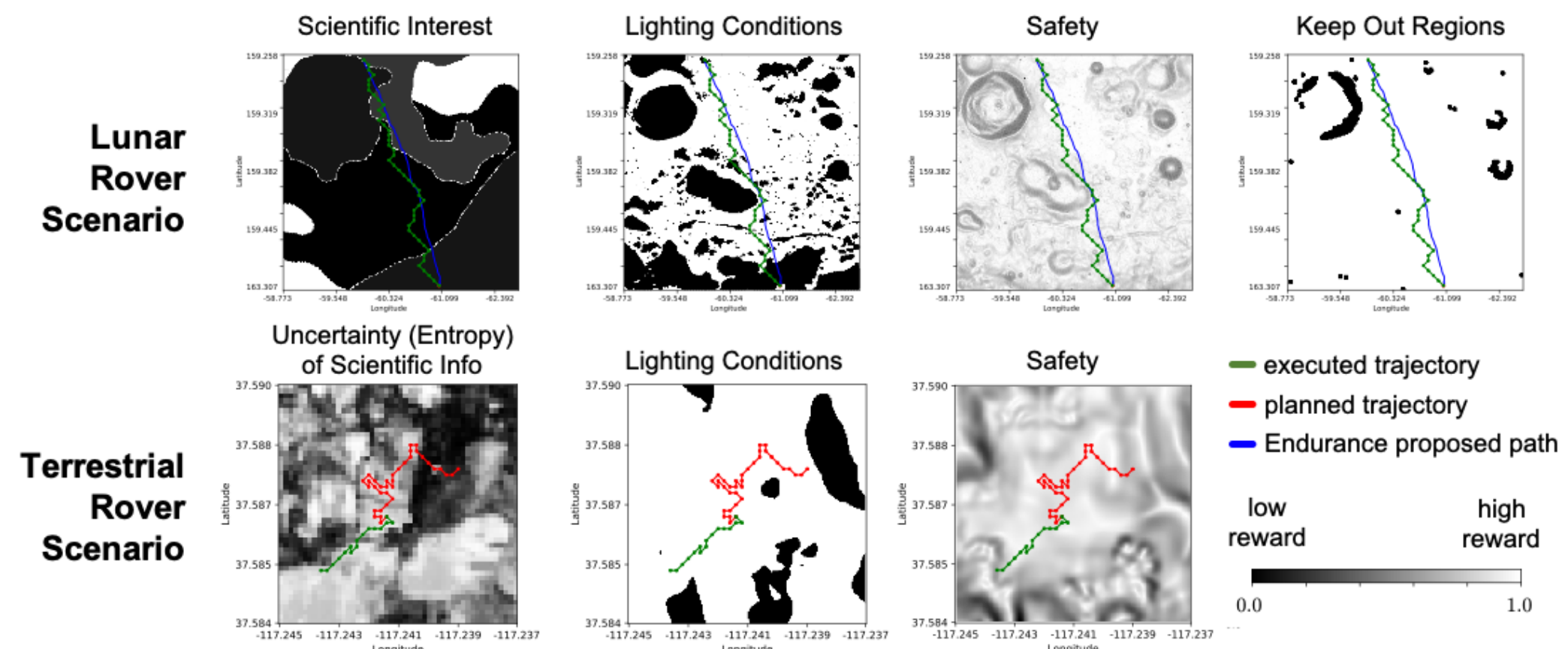


Fig. 2. Objective maps and example planned paths using the boundary intersection A* method for the lunar and terrestrial rover scenarios.

Our experiments demonstrate that the proposed explainable methods achieve equivalent or improved solution quality compared to the baselines (Fig. 3). The main advantage is explainability without sacrificing performance. These methods produce predictable actions and adhere more closely to the desired trade-off between objectives.

Lunar Rover Scenario										Terrestrial Rover Scenario					
Planning Method	Decomp. Method	Average % Interest Reduction ↑	Average In Sun % ↑	Average Safety % ↑	Average Keep In % ↑	Solution Quality (TOPSIS) ↑	P-value	Distance to λ ↓	Average % Entropy Reduction ↑	Average In Sun % ↑	Average Safety % ↑	Solution Quality (TOPSIS) ↑	P-value	Distance to λ ↓	
A*	EUCLID	78.765	89.551	92.186	99.683	0.869 ± 0.049	1.000	18.839	71.014	98.800	86.669	0.823 ± 0.024	0.862	19.979	
	BI	78.765	89.579	92.135	99.683	0.869 ± 0.049	0.593	18.819	71.013	98.760	86.636	0.823 ± 0.024	0.761	19.930	
	TCHBYCHEV	78.842	89.577	92.452	99.683	0.870 ± 0.049	0.893	18.826	70.230	97.280	85.264	0.814 ± 0.024	0.010	19.484	
	WEIGHTED	78.765	89.650	92.096	99.683	0.869 ± 0.049	-----	18.728	70.760	98.936	86.683	0.822 ± 0.023	-----	20.244	
MCTS	EUCLID	78.129	82.540	90.225	97.369	0.841 ± 0.046	0.075	20.362	67.388	97.832	83.793	0.798 ± 0.033	0.004	22.027	
	BI	77.910	82.352	90.210	97.075	0.840 ± 0.047	0.249	20.420	67.341	97.072	83.030	0.795 ± 0.029	0.083	21.507	
	TCHBYCHEV	77.300	80.553	89.531	97.490	0.832 ± 0.048	0.612	21.699	67.333	96.304	82.568	0.793 ± 0.034	0.332	21.070	
	WEIGHTED	77.480	82.217	90.341	97.369	0.838 ± 0.048	-----	21.032	65.946	97.776	83.063	0.789 ± 0.034	-----	23.163	
FIXED HORIZON	EUCLID	75.110	81.178	90.441	97.527	0.827 ± 0.061	0.715	23.730	68.062	96.584	84.638	0.801 ± 0.032	0.030	20.780	
	BI	75.040	81.316	90.658	97.725	0.828 ± 0.061	0.655	23.957	67.569	96.312	84.430	0.798 ± 0.034	0.351	20.997	
	TCHBYCHEV	76.504	81.095	89.950	97.413	0.832 ± 0.056	0.345	22.153	67.684	96.624	84.170	0.798 ± 0.029	0.265	21.059	
	WEIGHTED	75.046	80.275	90.654	97.725	0.824 ± 0.059	-----	24.717	67.385	96.432	84.127	0.796 ± 0.035	-----	21.191	
GREEDY	EUCLID	75.567	75.148	90.204	97.982	0.808 ± 0.056	-----	27.386	67.777	96.096	83.959	0.797 ± 0.036	0.632	20.572	
	BI	75.567	75.148	90.204	97.982	0.808 ± 0.056	-----	27.386	67.561	96.152	84.058	0.797 ± 0.035	0.664	20.752	
	TCHBYCHEV	75.567	75.148	90.204	97.982	0.808 ± 0.056	-----	27.386	66.947	95.904	83.707	0.793 ± 0.036	0.147	21.125	
	WEIGHTED	75.567	75.148	90.204	97.982	0.808 ± 0.056	-----	27.386	67.708	95.720	83.914	0.795 ± 0.039	-----	20.808	
ENDURANCE PATH		75.074	78.828	91.673	98.457	0.820 ± 0.056	-----	26.219							
RANDOM		76.147	73.041	88.406	96.38	0.804 ± 0.060	-----	24.974							

Fig. 3. Results for lunar and terrestrial rover scenarios.

Significance/Benefits to JPL and NASA: Long-distance autonomous driving technology is critical for upcoming lunar missions, including Endurance and Artemis activities using Lunar Terrain Vehicles (LTVs). This technology is essential for achieving mission goals while reducing operational costs by enabling longer autonomous drives and decreasing the need for frequent ground contact. Beyond the Moon, this work benefits future Mars missions, such as the Mars Science Helicopter and robotic elements of a human campaign. It is also applicable to any in-situ spacecraft designed for autonomous travel over significant distances to destinations like Titan, Europa, or Venus.

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Publications:

A. Breitfeld, A. Candela, S. Bhamidipati, T. Estlin, D. Wettergreen, "Multi-Objective Graph-Based Path Planning Using Decomposition Approaches for Exploration Rovers," submitted to *IEEE International Conference on Robotics and Automation (ICRA)*, Vienna, Austria, 2026.

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